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# Decoupled Third-order Fuzzy Sliding Model Control For Cart-Inverted Pendulum System

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**Abstract:** Cart Inverted Pendulum (CIP) system is a benchmark problem in nonlinear automatic control. It is commonly used to verify the robustness of any proposed nonlinear controller. CIP is mostly represented by two second order differential equations to avoid complexity due to the DC motor dynamics. This representation is not practical for the real CIP dynamics and might lead to instability. Therefore in this paper, two third-order differential equations were derived to combine the pendulum system and DC motor dynamics to have a more realistic mathematical model. Friction between the cart and rail was included in the system equations through a nonlinear friction model. To stabilize the obtained nonlinear electromechanical CIP model, a third-order Fuzzy Sliding Mode Controller (FSMC) was designed. The chattering of the control signal was eliminated using general bill shape membership functions for the Fuzzy controller. Simulation results proved the robustness of the proposed FSMC over Linear Quadratic Regulator Controller (LQRC). For instant, the overshoot in the cart position response was reduced by 300%.

Keywords: Sliding mode control, fuzzy Controller, cart-inverted pendulum system.

## 1. Introduction

Cart-Inverted Pendulum (CIP) system is considered as one of the most significant benchmark problems in dynamics. As a nonlinear unstable system, it affords many challenging tasks for control engineers. Furthermore, many practical restrictions present in controlling the system such as frictions, cart rail limitation and control signal saturation. All these limitations stimulate researchers to study the system extensively[1]. A lot of controllers have been proposed to stabilize the CIPS based on the linearized model [2,3].The closed loop stability is not considered and using linear control strategies might be insufficient, especially with existing of nonlinear friction and external disturbances. Thus, nonlinear control techniques are required to improve the system response and guarantee the stability.

Sliding Mode Control (SMC) is considered as a powerful tool controlling of nonlinear systems. As a robust control technique, SMC is an effective method dealing with system uncertainties and external disturbances. Another advantage of this approach is stability guarantee under external disturbances and parameters variations. Despite its effectiveness in many applications, SMC has a drawback of control signal chattering (high frequencies)[4].

On the other hand, Fuzzy Logic Control (FLC) is offering a good solution for systems with complicated mathematical model. Using the human experts, FLC provides a model free control scheme which has been applied to many engineering applications. The main disadvantage of FLC is the difficulties in guarantee the stability because of the linguistic expressions[5]. Recently, fuzzy sliding mode control (FSMC) has been introduced as a robust control method dealing with chattering problem. FSMC is working as a SMC with a boundary layer, where the FLC is forming the boundary layer to overcome high frequencies of the control signal, which is undesirable for many applications[6].

In literature, FSMC has been used in controlling of CIPS. Woo kim et al. [7] proposed single input fuzzy sliding mode controller SIFSMC, where the fuzzy rules were defined based on the sliding surface. Although, the results

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showed the validity of this controller for second- order nonlinear system, this class of controller is not appropriate for CIP. Using this controller, either the cart or the pendulum positions could be controlled, but not both. In order to overcome this problem, Ji-Chang Lo et al.[8] proposed a decoupled fuzzy sliding mode controller. In their work, the whole system was decoupled into two subsystems; each one has its control target. Decoupling function was designed to ensure that the control signal will control both of subsystems .Simulation results showed the effectiveness of the proposed controller. Advanced decoupling controller for nonlinear systems was introduced by Chih-Min Lin et al.[9] and Yoo et al.[10] Both showed improvements in system response in terms of speed convergence. Hung et al. [11] proposed a decoupled sliding mode based on neural networks to speed the system response. However, this control algorithm is convoluted and has techniqal complexity in practical implementation. With the aim of enhancing the system response, Yorgancioglu et al. [12] presented a time-varying sliding mode controller. This method improved the system output in terms of the integral of absolute error IAE. In [13], Tao et al. showed an adaptive fuzzy sliding surface controller to stabilize CIP. This controller objective is to reach two sliding surface disjointedly .The simulation results showed the validity for this scheme.

In[8, 12, 13], CIP was described as a forth order system (two nonlinear second order differential equations) and the system actuator (DC Motor) dynamics was neglected. However, from practical point of view, including the DC motor model will increase the system order to be a six order system (two third order differential equations),that makes such kind of controllers non-applicable for the real systems and might lead to the instability. This shows the need for designing a controller based on a model which includes the motor dynamics.

In this paper, a third order decouple FSMC is proposed for controlling the CIP .In this approach, the CIP is described as a sixth-order system, where the motor dynamics is added. In addition, nonlinear friction model is provided to describe the friction in the cart rail. The system model contains two subsystems (the cart and the pendulum) each is represented in a third order differential equation. In the FSMC, Two sliding surfaces are designed for both subsystems. Based on Lyapunov function, the control law is designed for the main subsystem to guarantee the stability and the system robustness. An Intermediate function is used to link the secondary and the main subsystems. This causes the control action to push both subsystems toward their stability. Finally, fuzzy controller is developed using a general bill membership functions to eliminate the control signal frequencies. Simulation results are disscused and compared with LQR technique.

## 2. Mathematical model

#### 2.1. CIPS model

CIPS has two degrees of freedom, X is the Cart displacement and  $\theta$  is the pendulum angle position, as shown in Figure 1. If the cart mass is donated by M, m is the pendulum mass, L is the length between the pivot and the pendulum center of gravity C.G, g is the acceleration of gravity, I is the pendulum mass moment of inertia with respect to its C.G.,  $F_{fr}$  is the friction force between the cart and the rail. q is the friction coefficient in the pendulum pivot. Based on D'Alembert's principle, The equations of motion are deduced to be:

$$F = (M+m)\ddot{X} + F_{fr} - m(L\ddot{\theta}\cos\theta - L\dot{\theta}^{2}\sin\theta)$$
(1)  
$$(I+mL^{2})\ddot{\theta} = mgL\sin\theta + mL\ddot{X}\cos\theta - q\dot{\theta}$$
(2)

For the friction Force  $F_{fr}$ , most of the former work, dealing with the CIPS, either has applied a viscous friction model(linear) or has neglected its effects. However, the friction phenomena encloses many terms such as Stribeck effects, static, Coulomb and viscous frictions. Thus, exponential friction model is chosen, to address all mentioned friction parts, as follows

$$F_{fr} = \begin{cases} \dot{X} \frac{F_s}{\dot{X}_d} \\ if \left| \dot{X} \right| \le \dot{X}_d \\ F_C + (F_S - F_C) e^{-\left| \dot{X} / V_S \right|^n} sgn(\dot{X}) + b\dot{X} \\ if \left| \dot{X} \right| > \dot{X}_d \end{cases}$$
(3)

where  $F_S$  is Static Friction force,  $F_C$  is Coulumb Friction force,  $X_d$  is the dead zone velocities,  $V_S$  is Stribeck velocity, n is form factor, b is the viscous friction coefficient.



Figure 1 Cart-inverted pendulum system



#### 2.2. DC motor model

Figure 2. illustrates the Dc motor Circuit, where  $V_a$  is the armature applied voltage (Control voltage),  $V_{emf}$  if the back EMF voltage,  $R_a$ ,  $L_a$  and i are the armature resistance, inductance and current, respectively.  $\omega$  is the DC Motor angular velocity,  $T_L$  is the Motor electromagnetic torque,  $T_J$  is motor inertia torque,  $T_B$  is the damping torque and  $T_L$  is the motor load torque. The motor equations are

$$V_a - V_{emf} = i R_a + L_a \frac{di}{dt} \tag{4}$$

$$V_{emf} = K_e \,\omega \tag{5}$$

 $K_e$  is the Back EMF constant. and

$$i = \frac{T_e}{K_t} \tag{6}$$

 $K_t$  is the motor torque constant. Since the motor and cart move with same speed, the following equation is obtained

$$\omega = \frac{\dot{X}}{r} \tag{7}$$

 $\boldsymbol{r}$  is the motor pulley diameter. The electromagnetic torque equation will be

$$T_e = T_J + T_B + T_L \tag{8}$$

where

$$T_J = J \dot{\omega} = J \frac{\ddot{X}}{r} \tag{9}$$

$$T_B = B\,\omega = B\,\frac{\dot{X}}{r} \tag{10}$$

$$T_L = F r \tag{11}$$

J is the motor rotor mass moment of inertia, B Motor rotor damping coefficient.



Figure 2 DC Motor circuit

### 2.3. Overall integrated model

Here, the overall system model will be derived and the expected model will be sixth order (two coupled third equations), where the motor applied voltage  $V_a$  is the system input. First, by substituting from (11),(10) and (9) in (8). And from(8) in (6) we get the current equation

$$i = \frac{T_e}{K_t} = \frac{J\frac{\dot{X}}{r} + B\frac{\dot{X}}{r} + [(M+m)\ddot{X} + F_{fr}]}{K_t}$$
$$\frac{-m(L\ddot{\theta}\cos\theta - L\dot{\theta}^2\sin\theta)]r}{K_t}$$
(12)

By taking the time derivative of the last equation, (13) is obtained

$$\frac{di}{dt} = \frac{\left[(M+m)r + \frac{J}{r}\right] + \frac{B}{r}\ddot{X} - mLr\cos\theta + mLr\ddot{\theta}\dot{\theta}\sin\theta}{K_t} + \frac{2mLr\ddot{\theta}\dot{\theta}\sin\theta + mLr\dot{\theta}^3\cos\theta + \dot{F}_{fr}}{K_t}$$
(13)

By substituting from (5),(12) and (13) in (4), we get

$$V_{a} = \left[\left[(M+m)r + \frac{J}{r}\right]\frac{L_{a}}{K_{t}}\right] + \left[\left[(M+m)r + \frac{J}{r}\right]\frac{R_{a}}{K_{t}}\right]\ddot{X} \\ + \left[\frac{BL_{a}}{rK_{t}}\right]\ddot{X} + \left[\left[\frac{BR_{a}}{rK_{t}}\right] + \left[\frac{K_{e}}{r}\right]\right]\dot{X} - \frac{rmLR_{a}}{K_{t}}\ddot{\theta}\cos\theta \\ + \frac{rmLR_{a}}{K_{t}}\dot{\theta}^{2}\sin\theta - \frac{rmLL_{a}}{K_{t}}\cos\theta + \frac{3rmLL_{a}}{K_{t}}\dot{\theta}\ddot{\theta}\sin\theta \\ + \frac{rmLL_{a}}{K_{t}}\dot{\theta}^{3}\cos\theta + \frac{R_{a}}{K_{t}}F_{fr} + \frac{L_{a}}{K_{t}}\dot{F}_{fr}$$
(14)

This equation is considered as the main overall equation, describing the system states with the applied voltage on DC motor as an input.

From equation (2)

$$\ddot{X} = -g \tan \theta + \frac{(I+mL^2)}{mL\cos\theta}\ddot{\theta} + \frac{q}{mL\cos\theta}\dot{\theta} \qquad (15)$$

$$\ddot{\theta} = \frac{mLg}{(I+mL^2)}\sin\theta + \frac{mL}{(I+mL^2)}\cos\theta\ddot{X} - \frac{q}{(I+mL^2)}\dot{\theta}$$
(16)

By taking the first derivative for(2) we get

$$(I + mL^{2})\ddot{\theta} = mgL\cos\theta\dot{\theta} + mL\ddot{X}\cos\theta$$
$$-mL\ddot{X}\sin\theta\dot{\theta} - q\ddot{\theta}$$
(17)

$$\ddot{X} = -g\dot{\theta} + \ddot{X}\,\tan\theta\,\dot{\theta} + \frac{q}{mL\,\cos\theta}\,\ddot{\theta} + \frac{(I+mL^2)}{mL\,\cos\theta}\,\ddot{\theta}$$
(18)

$$\ddot{\theta} = \frac{mgL}{(I+mL^2)}\cos\theta\dot{\theta} + \frac{mL}{(I+mL^2)}\ddot{X}\cos\theta - \frac{mL}{(I+mL^2)}\ddot{X}\sin\theta\dot{\theta} - \frac{q}{(I+mL^2)}\ddot{\theta}$$
(19)

Substituting from (15) and (18), with values of  $\ddot{X}$  and  $\ddot{X}$ , and substituting in equation (14) We get the following equation

$$\ddot{\theta} = \frac{f_3 \dot{\theta} - f_4 \tan^2 \theta \dot{\theta} + f_5 \frac{\tan \theta \dot{\theta} \ddot{\theta}}{\cos \theta} + f_6 \frac{\tan \theta \dot{\theta}^2}{\cos \theta} + f_7 \frac{\ddot{\theta}}{\cos \theta}}{[f_1 \cos \theta - \frac{f_2}{\cos \theta}]}$$

$$\frac{-f_8 \tan \theta + f_9 \frac{\dot{\theta}}{\cos \theta} + f_{10} \dot{X} - f_{11} \ddot{\theta} \cos \theta}{[f_1 \cos \theta - \frac{f_2}{\cos \theta}]}$$

$$\frac{+f_{12} \dot{\theta}^2 \sin \theta + f_{13} \dot{\theta} \ddot{\theta} \sin \theta + f_{14} \dot{\theta}^3 \cos \theta + f_{15} F_{fr}}{[f_1 \cos \theta - \frac{f_2}{\cos \theta}]}$$

$$\frac{+f_{16fr}}{[f_1 \cos \theta - \frac{f_2}{\cos \theta}]} - \frac{1}{[f_1 \cos \theta - \frac{f_2}{\cos \theta}]} V_a \qquad (20)$$

#### Where the values of constants $f_{1\rightarrow 16}$ are

$$\begin{split} f_1 &= \frac{r \, m \, L \, L_a}{K_t}, \, f_2 = \frac{\left[(M+m)r + \frac{J}{r}\right] \ L_a \ \left[I + mL^2\right]}{K_t \, m \, L}, \\ f_3 &= \frac{\left[(M+m)r + \frac{J}{r}\right] \ L_a \, g}{K_t \, m \, L}, \, f_4 = \frac{\left[(M+m)r + \frac{J}{r}\right] L_a g}{K_t \, m \, L}, \\ f_5 &= \frac{\left[(M+m)r + \frac{J}{r}\right] \ L_a \ \left[I + mL^2\right]}{K_t \, m \, L}, \, f_6 = \frac{\left[(M+m)r + \frac{J}{r}\right] \ L_a \ q}{K_t \, m \, L}, \\ f_7 &= \frac{\left[(M+m)r + \frac{J}{r}\right] L_a q + \left[\left[(M+m)r + \frac{J}{r}\right] R_a + \left[\frac{B \, L_a}{K_t \, m \, L}\right]\right](I + mL^2)}{K_t \, m \, L} \\ f_8 &= \left[\left[(M + m) \, r + \frac{J}{r}\right] \frac{R_a}{R_t} + \left[\frac{B \, L_a}{R_t}\right]\right] g, \\ f_9 &= \frac{\left[\left[(M+m)r + \frac{J}{r}\right] R_a + \left[\frac{B \, L_a}{R_t}\right]\right] q}{K_t \, m \, L} \\ f_{10} &= \left[\left[\frac{B \, R_a}{r \, K_t}\right] + \left[\frac{K_e}{r}\right]\right], \, f_{11} &= \frac{rmLR_a}{K_t}, \, f_{12} &= \frac{rmLR_a}{K_t} \\ f_{13} &= \frac{3rmLL_a}{K_t}, \, f_{14} &= \frac{rmLL_a}{K_t}, \, f_{15} &= \frac{R_a}{K_t}, \, f_{16} &= \frac{L_a}{K_t} \end{split}$$

equation(20) is the first overall system equation which will be rewritten in the form:

$$\ddot{\theta} = \alpha_1(\dot{\theta}, \theta, \dot{X}, X) + \beta_1(\dot{\theta}, \theta, \dot{X}, X) V_a$$
 (21)

where

$$\alpha_{1} = \frac{f_{3}\dot{\theta} - f_{4}\tan^{2}\theta\dot{\theta} + f_{5} \frac{\tan\theta\dot{\theta}\ddot{\theta}}{\cos\theta} + f_{6} \frac{\tan\theta\dot{\theta}^{2}}{\cos\theta} + f_{7}\frac{\ddot{\theta}}{\cos\theta}}{[f_{1}\cos\theta - \frac{f_{2}}{\cos\theta}]}$$

$$\frac{-f_{8}\tan\theta + f_{9}\frac{\dot{\theta}}{\cos\theta} + f_{10}\dot{X} - f_{11}\ddot{\theta}\cos\theta}{[f_{1}\cos\theta - \frac{f_{2}}{\cos\theta}]}$$

$$\frac{+f_{12}\dot{\theta}^{2}\sin\theta + f_{13}\dot{\theta}\ddot{\theta}\sin\theta + f_{14}\dot{\theta}^{3}\cos\theta + f_{15}F_{fr}}{[f_{1}\cos\theta - \frac{f_{2}}{\cos\theta}]}$$

$$\frac{+f_{16fr}}{[f_{1}\cos\theta - \frac{f_{2}}{\cos\theta}]}$$
(22)

$$\beta_1 = -\frac{1}{\left[f_1 \cos \theta - \frac{f_2}{\cos \theta}\right]} \tag{23}$$

To get the second overall system equation, substituting from (19) and (16) with values of  $\ddot{\theta}$  and  $\ddot{\theta}$  in equation (14)

$$\ddot{X} = \frac{-f'_{3} \cos^{2}\theta \dot{\theta} + f'_{4} \dot{X} \sin \theta \cos \theta \dot{\theta} + f'_{5} \cos \theta \sin \theta}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ + \frac{f'_{6} \cos^{2}\theta \ddot{X} + f'_{7} \cos \theta \dot{\theta} + f'_{8} \ddot{X} + f'_{9} \dot{X}}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ + \frac{f'_{10} \dot{\theta}^{2} \sin \theta + f'_{11} \dot{\theta} \sin^{2}\theta + f'_{12} \dot{\theta} \sin \theta \cos \theta \ddot{X}}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ + \frac{f'_{13} \dot{\theta}^{3} \cos \theta + f'_{14} F_{fr} + f'_{15} \dot{F}_{fr}}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ - \frac{1}{[f'_{2} \cos^{2}\theta - f'_{1}]} V_{a}$$
(24)

Where the values of constants  $f'_{1\rightarrow15}$  are

$$\begin{split} f_1' &= \frac{\left[ (M+m)r + \frac{1}{r} \right] L_a}{K_t}, f_2' = \frac{r m^2 L^2 L_a}{(I+mL^2) K_t} \\ f_3' &= \frac{r m^2 L^2 L_a q}{(I+mL^2) K_t}, f_4' = \frac{r m^2 L^2 L_a q}{(I+mL^2) K_t} \\ f_5' &= \frac{r m^2 L^2 L_a q q}{(I+mL^2)^2 K_t} - \frac{r m^2 L^2 R_a q}{(I+mL^2) K_t} \\ f_6' &= \frac{r m^2 L^2 L_a q}{(I+mL^2)^2 K_t} - \frac{r m^2 L^2 R_a}{(I+mL^2) K_t} \\ f_7' &= -\frac{r m L L_a q^2}{(I+mL^2)^2 K_t} + \frac{r m L R_a q}{(I+mL^2) K_t} \\ f_8' &= \left[ \left[ (M+m) r + \frac{J}{r} \right] \frac{R_a}{K_t} + \left[ \frac{B L_a}{r K_t} \right] \right] \\ f_{10}' &= \frac{r m L R_a}{K_t} - \frac{3r m L L_a q}{(I+mL^2) K_t} \\ f_{11}' &= \frac{3r m^2 L^2 L_a q}{(I+mL^2) K_t}, f_{12}' &= \frac{3r m^2 L^2 L_a}{(I+mL^2) K_t} \\ f_{13}' &= \frac{r m L L_a}{K_t}, f_{14}' &= \frac{R_a}{K_t}, f_{15}' &= \frac{L_a}{K_t} \end{split}$$

equation(24) is the second overall system equation which will be rewritten in the form:

$$\ddot{X} = \alpha_2(\dot{\theta}, \theta, \dot{X}, X) + \beta_2(\dot{\theta}, \theta, \dot{X}, X)V_a$$
(25)

$$\alpha_{2} = \frac{-f'_{3} \cos^{2}\theta \dot{\theta} + f'_{4} \ddot{X} \sin \theta \cos \theta \dot{\theta} + f'_{5} \cos \theta \sin \theta}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ \frac{+f'_{6} \cos^{2}\theta \ddot{X} + f'_{7} \cos \theta \dot{\theta} + f'_{8} \ddot{X} + f'_{9} \dot{X}}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ \frac{+f'_{10} \dot{\theta}^{2} \sin \theta + f'_{11} \dot{\theta} \sin^{2}\theta + f'_{12} \dot{\theta} \sin \theta \cos \theta \ddot{X}}{[f'_{2} \cos^{2}\theta - f'_{1}]} \\ \frac{+f'_{13} \dot{\theta}^{3} \cos \theta + f'_{14} F_{fr} + f'_{15} \dot{F}_{fr}}{[f'_{2} \cos^{2}\theta - f'_{1}]}$$
(26)

$$\beta_2 = -\frac{1}{[f'_2 \cos^2\theta - f'_1]} V_a \tag{27}$$

#### 3. Controller design

# 3.1. FSMC design

In the following section, the FSMC is presented based on the derived integrated model. First, the third-order SMC is obtained using the sliding surface concept. Next, FLC is design and applied jointly with SMC to overcome the control signal chattering. From the system equations (21) and (25), and if  $D_1(t)$  and  $D_2(t)$  are bounded external disturbances, the entire system model will have the following form

$$\ddot{\theta} = \alpha_1 + \beta_1 V_a + D_1 \tag{28}$$

$$\ddot{X} = \alpha_2 + \beta_2 V_a + D_2 \tag{29}$$

where  $\alpha_1$  and  $\beta_1$  are nonlinear functions of the system states  $\theta$ , X,  $\dot{\theta}$ ,  $\dot{X}$  and  $\ddot{\theta}$ .  $\alpha_2$  and  $\beta_2$  are functions of  $\theta$ , X,  $\dot{\theta}$ ,  $\dot{X}$  and  $\ddot{X}$ .

For CIPS, the system states are $\theta$ ,  $\theta$ , X, X and X. There SMC will be used in the third-order form. The control law is designed based on the sliding surface where the system states is moved from any general position towards the sliding surface and keep sliding till the equilibrium point. The general equation for the sliding surface S is [7]

$$S(x,t) = (\frac{d}{dt} + C)^{(n-1)}x$$
 (30)

where x is the system sate, n is the system order and C is a constant value. In this case(CIPS) the system consists of two third order subsystems. So that, two sliding surface  $S_2$  for the pendulum subsystem and  $S_1$  for the Cart subsystem, are considered where

$$S_1 = C_1^2 \theta + 2C_1 \dot{\theta} + \ddot{\theta} \tag{31}$$

$$S_2 = C_2^2 X + 2C_2 \dot{X} + \ddot{X}$$
(32)

where  $C_1$  and  $C_2$  are a positive constants. Sliding surfaces S1 andS2 are constructed based on the values of  $C_1$  and  $C_2$ . Thus, the system response is highly dependent on Values of C1 and C2, appropriate values of these constant will achieve the desired system response.

Based on the sliding surfaces, the control law will be generated. Since only one control action is available, the Pendulum angle will be considered as primary control target and the cart position is the secondary one. First, the controller is designed to achieve the primary target where  $S_1 = 0$ . Then, an intermediate function is used to link between the secondary and primary targets. This function will achieve the cart subsystem stability if the pendulum stability is reached.

Considering Lyapunov function

$$V = \frac{1}{2}S_1^{\ 2} \tag{33}$$

As it is known from Lyapunov theorem, if V is negative, that means the system will be moving to the sliding surface and remains sliding till the stability is achieved

$$\dot{V} = \dot{S}_1 S_1 \tag{34}$$

$$\dot{V} = S_1(C_1{}^2\dot{\theta} + 2C_1\ddot{\theta} + \ddot{\theta})$$

from equation 3.1

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$$\dot{V} = S_1 (C_1^2 \dot{\theta} + 2C_1 \ddot{\theta} + \alpha_1 + \beta_1 V_a + D_1)$$

If the control signal  $V_a$  has the following form,  $\dot{V}$  will be negative:

$$V_a = \hat{V}_a - K \operatorname{sgn}(S_1 \beta_1), \quad K > \frac{D_1}{|\beta_1|}$$
 (35)

where

$$\hat{V}_a = \frac{-C_1^2 \dot{\theta} - 2C_1 \ddot{\theta} - \alpha_1}{\beta_1}$$

This form of the control signal guarantee the stability For CIPS, the system states  $\operatorname{are}\theta$ ,  $\dot{\theta}$ ,  $\ddot{X}$ ,  $\dot{X}$  and  $\ddot{X}$ . Therefore, for the pendulum subsystem since  $\dot{V}$  is kept negative. In order to ensure for the whole system stability, the intermediate link function is introduced as follows:

First, the first sliding surface will be reformed to be

$$S_1 = C_1^2(\theta - Z) + 2C_1\dot{\theta} + \ddot{\theta}$$
 (36)

based on the new sliding surface, the control target changed from  $\begin{bmatrix} \theta & \dot{\theta} & \ddot{\theta} \end{bmatrix}^T = \begin{bmatrix} 0 & 0 & 0 \end{bmatrix}^T$  to be  $\begin{bmatrix} \theta & \dot{\theta} & \ddot{\theta} \end{bmatrix}^T = \begin{bmatrix} Z & 0 & 0 \end{bmatrix}^T$ . Where Z is the intermediary function which is a function of  $S_2$ . The objective  $S_2 = 0$  is embedded in the main control target through the variable which is defined to be

$$Z = sat(\frac{S_2}{\phi_z})Z_U \tag{37}$$

 $Z_U$  is the upper limit of the function,  $\phi_z$  is the function boundary layer. At any general condition, the controller will move the system states  $\dot{\theta}$  and  $\ddot{\theta}$  to zero. consequently, the intermediate function Z converges to zero. In that case and from equation (36) and (37), it could be found that both of  $S_1$  and  $S_2$  will be decaying to zero [14].

The control action Va, as it is shown in equation (35), has a high-frequencies switching because of the Sgn function. To overcome this problem, a bounder layer will be formed using a simple fuzzy controller. Such a kind of controller will be capable of reject all the high-frequencies and solve the chattering problem. The controller equation will be modified to be

$$V_a = \hat{V}_a - K_{fuzzy} \tag{38}$$

 $K_{fuzzy}$  is the fuzzy controller output.

Nine general bill type membership functions have been chosen for the input  $(S_1B_1)$  and output  $(K_{fuzzy})$ . This type will generate a smooth curve to reject any undesired frequencies. As it is illustrated in Figure 3. and Figure 4., ZR is zero, NS is negative small, NB is negative big, NVB is negative very big, PS is positive small, PM positive medium and, PB is Positive big, PVB is positive very big.  $\phi$  is the upper bound of the boundary layer which is an indicators for the width of boundary layers. Thus, choosing suitable values for  $\phi$  will reject any undesirable frequencies. K is the fuzzy output saturation  $\phi$  and should be selected according to the DC motor limits



The fuzzy surface is formed using the following IF-THEN rules

IF  $(S_1B_1)$  is NVB THEN  $(K_{fuzzy})$  is PVBIF  $(S_1B_1)$  is NB THEN  $(K_{fuzzy})$  is PBIF  $(S_1B_1)$  is NM THEN  $(K_{fuzzy})$  is PMIF  $(S_1B_1)$  is NS THEN  $(K_{fuzzy})$  is PSIF  $(S_1B_1)$  is PS THEN  $(K_{fuzzy})$  is NSIF  $(S_1B_1)$  is PS THEN  $(K_{fuzzy})$  is NSIF  $(S_1B_1)$  is PM THEN  $(K_{fuzzy})$  is NMIF  $(S_1B_1)$  is PB THEN  $(K_{fuzzy})$  is NMIF  $(S_1B_1)$  is PB THEN  $(K_{fuzzy})$  is NBIF  $(S_1B_1)$  is PB THEN  $(K_{fuzzy})$  is NBIF  $(S_1B_1)$  is PVB THEN  $(K_{fuzzy})$  is NVB

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The fuzzy controller output is shown in Figure 5.



Figure 3 Input membership functions



Figure 4 Output membership function



Figure 5 Fuzzy controller surface

# 3.2. LQRC design

LQR control technique is widely used in controlling linear system. It has been applied to stabilize CIP after linearzing the state-space system equations around the pendulum angle stability position[2] [3]. In this case, the system states are defined to be  $x_1 = \theta$ ,  $x_2 = \dot{\theta}$ ,  $x_3 = \ddot{\theta}$ ,  $x_4 = X$ ,  $x_5 = \dot{X}$  and  $x_6 = \ddot{X}$ . By approximating third order differental equations (2.21)and (2.25) around the equilibrium point where  $x = [0 \ 0 \ 0 \ 0 \ 0 \ 0 \ 0]^T$ , the overall system linear equation is obtained. The general state-sapce form is

$$\dot{x} = Ax + Bu \tag{39}$$

where x is state matrix (1x6), u control signal matrix (1x1), A state parameters matrix (6x6), B control signal parameters matrix (1x6). Since only one control action(DC motor voltage) is available,  $u = V_a$ .

The equivalent state-space linearized system equation is

$$\begin{pmatrix} \dot{\theta} \\ \ddot{\theta} \\ \ddot{\theta} \\ \dot{X} \\ \dot{X} \\ \dot{X} \\ \ddot{X} \end{pmatrix} = \begin{pmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 \\ \frac{-f_8}{f_1 - f_2} & \frac{f_3 + f_9}{f_1 - f_2} & \frac{f_7 - f_{11}}{f_1 - f_2} & 0 & \frac{f_{10}}{f_1 - f_2} & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ \frac{f_5'}{f_2' - f_1'} & \frac{f_7' - f_3'}{f_2' - f_1'} & 0 & 0 & \frac{f_9'}{f_2' - f_1'} & \frac{f_6' + f_8'}{f_2' - f_1} \end{pmatrix} \begin{pmatrix} \theta \\ \dot{\theta} \\ \ddot{X} \\ \dot{X} \\ \dot{X} \\ \dot{X} \end{pmatrix} \\ + \begin{pmatrix} 0 \\ 0 \\ \frac{1}{f_2 - f_1} \\ 0 \\ 0 \\ \frac{1}{f_2 - f_1'} \end{pmatrix} V_a$$
(40)

From LQR theory, the following sate feedback control law is applied.

$$u = V_a = -Kx \tag{41}$$

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Where K is the optimal feedback gain matrix required to get a minimum performance index J

$$J = \int_0^\infty (x^T Q x + u^T R u) \tag{42}$$

Where Q and R are a real symmetric matrices which are chosen by the designer. The gain matrix K is calculated by solving Reduced-matrix Riccati equation (43), after obtaining matrix P.

$$A^{T}P + PA - PBR^{-1}B^{T}P + Q = 0 (43)$$

where P is an intermediate matrix used to calculate the gain matrix K [15]

$$K = R^{-1}B^T P \tag{44}$$

The controller parameter Q should be carefully chosen based on the states priority. Form a control point of view, the pendulum angle  $\theta$  is much more important than the cart position X. Therefore, a bigger value should be chosen for the angle  $\theta$  element in Q matrix. Selection of R matrix value depends on the control signal constrains. Based on the values of Q and R, the feedback gain matrix K is obtained.

## 4. Simulation results

In this section, the CIP system is simulated based on equations (21) and (25). A comparison study has been carried out between the proposed FSMC and LQRC technique to stabilize the system. The cart rail limits is assumed to be  $X = \pm 0.35m$  and the DC motor maximum voltage is  $V_a = \pm 24V$ . A nonlinear friction force is considered between the cart and the rail according to equation (3). However, in controller design, this force is a source of uncertainty and disturbance affecting the cart motion. All CIP parameters used in this simulation are listed in Table 1. Simulation has been done using Matlab Simulink.

Table 1	System	parameters
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parameter value		parameter value		
M	$0.882 \ kg$	$K_t$	$0.00767 \ N.m/A$	
m	0.23~kg	$L_a$	$0.18 \mathrm{x} 10^{-3} \mathrm{H}$	
L	$0.3302 \ m$	$R_a$	2.6 Ohm	
I	7.88x 10 <sup>-3</sup> kg.m <sup>2</sup>	r	$6.35 \text{x} 10^{-3} \text{m}$	
g	9.81 $m/s^2$	$K_e$	$0.00767 \ V.s/rad$	
q	$0.0001 \ N.s/rad$	J	3.9x10 <sup>-7</sup> kg.m <sup>2</sup>	
$F_S$	0.2 N	$F_C$	0.18 N	
$\dot{X}_d$	$0.01 \ m/s$	$V_S$	$0.1\ m/s$	
n	4	b	$2.5 \ N.s/m$	
В	8x10 <sup>-7</sup> N.m.s/rad		-	

Controller parameters must be carefully selected where all system limits should be considered. All FSMC and LQRC parameters are shown in Table 2 and Table 3, respectively.

 Table 2
 FSMC parameters

parameter	$C_1$	$C_2$	$\phi$	$\phi_z$	K
Value	4.6	2.2	$1 \times 10^{4}$	19	15

Table 3 LQRC parameters

parameter	Value
Q	diag [25 1 1 4 1 1]
R	0.02
K	[234.16 68.04 5.79 -14.14 -25.50 2.50]

Using the initial conditions  $\theta = 0.1, \dot{\theta} = 0, \ddot{\theta} =$ 0, X = 0,  $\dot{X} = 0$  and  $\ddot{X} = 0$ , the response of the pendulum angular position using the proposed FSMC and LQRC are shown in figure 6.In spite of the nonlinear friction force, The proposed controller could reach the stability position faster than LQRC which has some oscillations around the stability position. figure 7. shows the cart position response. A reduced value of overshoot using FSMC (around 0.1 m) is observed in comparing with LQRC where the overshoot has reached 0.3 m. The LQRC response has continued to show oscillations around the zero position. The control signal response is shown in figure 8. The full stability is achieved by FSMC whereas LQRC response shows oscillations near the zero voltage. Thanks for the Fuzzy controller, all high-frequencies are eliminated in the FSMC response.



Figure 6 Pendulum angle response



Figure 7 Cart position response



Figure 8 Control signal response

Moreover, to verify the robustness of the proposed FSMC, an external disturbance, with value of 0.05 rad (50% of the initial unstable position) and one second duration, has been applied after 20s. The pendulum and cart responses are shown in figure 9. and figure 10,,respectively. The pendulum angular response shows a faster response with FSMC and able to reject the disturbance more efficiently than LQRC. The maximum overshoot has been reduced by %100 using the FSMC. The cart response shows the robustness of the FSMC over the LQRC where the maximum overshoot increased by %300. Furthermore, in the LQRC, the cart has exceeded the rail limits.

The simulation results revealed that FSMC provides faster response and better robustness comparing to LQRC under similar practical constraints.

#### 5. Conclusion

In this paper, two third-order differential equations have been developed to combine the CIP with its DC motor.



Figure 9 Pendulum angular position response under disturbance



Figure 10 Cart response under disturbance

Nonlinear friction model is added to describe the friction between the cart and the rail. A third-order FSMC is proposed to stabilize the derived CIP model. This controller guarantees the overall system stability. Chattering and all undesirable frequencies in the control signal has been eliminated by forming a boundary layer using a fuzzy controller. A third-order LQRC is also presented and compared with the proposed FSMC. Despite the system constrains and external disturbance (50% of the initial condition), simulation results of FSMC show its effectiveness and robustness over the LQRC. Using the proposed FSMC instead of LQRC has decreased the cart response overshoot by 300% and the pendulum response by %100.

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